Sub-Arcsecond Performance of the ST5000 Star Tracker on a Balloon-Borne Platform

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ANITA

BLAST

SBI

This talk is in three parts:
About balloon-borne telescopes
Pointing with ST5000 star tracker
Next steps for the ST5000

PART ONE: NASA's Balloon Program







CREAM Trajectory, 19 Days 2009



- BPO flies payloads recommended (& funded) by NASA programs like APRA.
- Zero-pressure balloons, up to 8000 lb payloads, daily 10% vent/ballast cycles.
- Durations: 3-4 weeks from Antarctica.
- Altitudes: I 20,000 ft, above 99.5% of the atm.
- Worldwide launches (e.g., from Palestine, Ft Sumner, Kiruna, Alice Springs and Antarctica).





| | Conventional | LDB | ULDB* |
|--|---|---|--|
| Duration | ~ 2 days | 40+ days | Up to 100 days |
| Flight Opportunities | ~16 per year | 3-6 per year | 1 per year |
| Suspended Capacity | 1650-8000 lbs | | 6000 lbs |
| Float Altitude | Up to 160,000 ft | | Up to 110,000 ft |
| Flight Support Systems All NASA Flight Support Systems are highly reliable proven systems | CIP •Line of Sight •300 kbps direct return | SIP •Over the Horizon •100 kbps TDRSS downlink | CDM/SIP •Over the Horizon •100 kbps TDRSS downlink |
| Launch Locations Operations Costs per flight, excluding Instrument | Fort Sumner, NM; Palestine, Texas; Alice Springs, Australia ROM \$ 100-250 K | Antarctica; Kiruna, Sweden; Alice Springs, Australia; ROM \$ 250- 500 K ROM \$ 500- 1000 K | |











| Balloon Size | Maximum Suspended Weight (lbs) | Float Altitude (ft) | |
|--------------|-----------------------------------|---------------------|--|
| 59.84 MCF | 1650 | 160,000 | |
| 36.73 MCF | 8000 | 120,000 | |
| 39.97 MCF | 6000 | 127,000 | |
| 28.47 MCF | 6500 | 119,000 | |
| 11.82 MCF | 7450 | 98,000 | |
| 11.82 MCF | 2875 | 116,000 | |
| 4.0 MCF | 3500 | 96,000 | |



QUESTION: What is the environment like in the stratosphere at 120,000 ft (35 km)?

Specifically:

- How good is the seeing?
- How stable is the photometry?
- What is the telluric spectrum from 120,000 ft?
- What is the infrared background?
- What is the daytime background?



Question: What is the atmospheric seeing at 120,000 ft?

Natural Seeing

A Critical Feasibility Question: Will speckles generated by aberrations of free-atmospheric origin permit 10⁻⁹ contrast limit, assuming a perfect

coronagraph?



Anisotropic Inhomogeneities



| Seeing parameter | Balloon-borne (35 km alt.) | Ground- based |
|-------------------------------|-------------------------------|------------------|
| Fried r ₀ | 41 m | 0.2 m |
| Inner scale I ₀ | 2.4 m | 0.006 m |
| Outer scale Λ_0 | 44 m | 27 m |

Gurvich & Chunchuzov (2003) *JGR*, Gurvich & Brekhovskikh (2001) *Waves in Radom Media*



C_n² vs. Altitude

Pin Chen, et al.

From a presentation by Pin Chen et al. 2009, KISS workshop on Innovative Approaches to Exoplanet Spectra.

< <u>http://www.kiss.caltech.edu/workshops/exoplanet2009/</u>>

Question: What is the atmospheric seeing at 120,000 ft?



Consider the SUNRISE mission, which launched from Kiruna (Sweden) in 2009.

During its 7-day mission it acquired tracked its target (the Sun) with rms errors of a few tens of milli-arcseconds.

It acquired the sharpest images to date of the Sun.



The SUNRISE Mission's Shack-Hartmann Experiment

T. Berkefeld et al. 106 Field stop Coll1 LLA Coll2 Reimager CCD The Wave-Front Correction System for the Sunrise T. Berkefeld · W. Schmidt · D. Soltau · A. Bell · H.P. Doerr · B. Feger · R. Friedlein · K. Gerber · F. Heidecke · T. Kentischer · O. v. d. Lühe · M. Sigwarth · E. Wälde · P. Barthol · W. Deutsch · A. Gandorfer · D. Germerott · B. Grauf · R. Meller · A. Álvarez-Herrero · M. Knölker · V. Martínez Pillet · S.K. Solanki · A.M. Title Received: 8 June 2010 / Accepted: 4 November 2010 / Published online: 8 December 2010

Figure 3 Scheme of the wave-front sensor of Sunrise. The field stop coincides with a focal plane of the telescope.

Figure 4 Illumination pattern of the CWS lenslet array. The image of the 1 m entrance pupil provides a homogeneous illumination of the six peripheral micro-lenses, except for the (small) influence of the spiders. The central lenslet is obscured by the secondary mirror and is not used.

QUESTION: How good is the seeing at 120,000 ft?

Solar Phys (2011) 268: 103-123

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THE SUNRISE BALLOON-BORNE OBSERVATORY

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The purpose of a Shack-Hartmann array

is to split up an aperture into smaller sub-

Balloon-Borne Solar Observatory

caused by the atmosphere.

The 2009 SUNRISE flight quantitatively measured the effects of seeing. Result: you cannot tell that you are not in **space** from the Correlated Wavefront Sensor results.

apertures. Relative motion between the images formed by the Shack-Hartmann lenslets indicates wavefront distortion





The SUNRISE Mission's Shack-Hartmann Experiment

:0"

The Sunrise Mission

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Abstract The first science flight of the balloon-borne Sunrise telescope took place in June 2009 from ESRANGE (near Kiruna/Sweden) to Somerset Island in northern Canada. We



QUESTION: How good is the seeing at 120,000 ft?

The 2009 SUNRISE flight quantitatively measured the effects of seeing. Result: **you cannot tell that you are not in space** from the Correlated Wavefront Sensor results.



Diffraction-Limited Performance from 120,000 ft

| | lm | I.5 m | 2 m |
|---------|--------|--------|--------|
| 0.25 µm | 0.063" | 0.042" | 0.031" |
| 0.5 µm | 0.126" | 0.084" | 0.063" |
| Ιμm | 0.252" | 0.168" | 0.126" |

A one meter telescope in the stratosphere will provide a 0.125 arcsec point spread function (PSF) in visible wavelengths.

What about photometry?

Wavelength

What is the Amplitude of Scintillation at 120,000 ft?

Robert et al.

Vol. 25, No. 2/February 2008/J. Opt. Soc. Am. A 379

Retrieving parameters of the anisotropic refractive index fluctuations spectrum in the stratosphere from balloon-borne observations of stellar scintillation

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Scintillation effects are not negligible in the stratosphere. We present a model based on a 3D model of anisotropic and isotropic refractive index fluctuations spectra that predicts scintillation rates within the so-called small perturbation approximation. Atmospheric observations of stellar scintillation made from the AMON-RA (AMON, Absorption par les Minoritaires Ozone et NO_x ; RA, rapid) balloon-borne spectrometer allows us to remotely probe wave-turbulence characteristics in the stratosphere. Data reduction from these observations brings out values of the inner scale of the anisotropic spectrum. We find metric values of the inner scale that are compatible with space-based measurements. We find a major contribution of the anisotropic spectrum relative to the isotropic contribution. When the sight line plunges into the atmosphere, strong scintillation occurs as well as coupled chromatic refraction effects. © 2008 Optical Society of America

Normalize

OCIS codes: 010.1300, 010.1330, 010.1290, 280.0280, 120.6200.

Short answer:

At astronomical elevation angles $(5^{\circ} - 75^{\circ})$ from z = 35 km, scintillation will be a negligible noise source.

Robert et al. (2008) fit a scintillation model to balloon-borne observations (z = 29.2 km). Their model accurately predicts scintillation (except at chords corresponding to elevation angles below 3°!).



Eliot Young • ST5000 Balloon Test Flight

What is the Daytime Sky Background at 120,000 ft?

Sky Background model:

- Decreases with λ^{-4}
- Decreases by 2x for every 5km increase in altitude
- Worse at angles 45° from sun or less





On May 5, 2011 we flew an ST5000 star tracker to z = 35 km. This image from 20 min before sunrise shows significant background in visible wavelengths (about 500 nm).

Short answer: Daytime background is significant. About 100x less than from Mauna Kea. Not good for UV/Visible targets.



What is the Mid-Infrared Environment at 120,000 ft?



The telluric methane and water lines are almost completely absent and transmission exceeds 70% within the CO₂ band.

Considerably lower downwelling radiance than at lower altitudes, enabling longer observing for dim objects.



What is the Thermal IR Environment at 120,000 ft?



At 36 km, transmission is nearly 100% throughout the TIR and downwelling radiance is several times less than at 40K', and orders of magnitude less than at Mauna Kea. O_3 and CO_2 minimally absorb. Similar measurement advantages to Mid-IR



Summary of the near-space environment at I 20,000 ft:

- \diamond Imaging: diffraction limited at $\lambda < I \mu m$.
- Photometry: virtually no scintillation.
- ♦ Daytime: A problem at λ < I µm.
- ♦ Low telluric absorptions.
- ♦ Low IR backgrounds.

Balloon-borne telescopes have an important role at $\lambda < I \mu m$. This is a regime that is only accessible to the Hubble Space Telescope. It not covered by ground-based adaptive optics systems.

PART TWO: THE ST5000 TEST FLIGHT

Question: Why fly a star tracker?

Answer: If the stratosphere supports DIFFRACTION LIMITED imaging, balloon platforms need to be pointed at a fraction of the diffraction limit.

This presents two challenges:

- Getting the pointing error, and
- Implementing the corrections.

The Star Tracker 5000 might be a reasonable way to address the first problem. The ST5000 was developed at the University of Wisconsin. It has become the standard star tracker for NASA's sounding rocket program.

Cost: less than \$100K per unit.

The ST5000 Star Tracker

- Developed by the University of Wisconsin-Madison/Space Astronomy Laboratory
- Used by NSROC/Flown on many sounding rockets
- Achieves Lost in Space solution in 1 s.
- I0 Hz attitude updates
- Weight: 9 lb; Power:
 I 2 W
- Accuracy of pointing solution: 0.5"
- 50 mm F/0.95 lens
- 7°x5° FOV, 35" per pixel





The ST5000 Star Tracker

- Question: ST5000 pixels are ~35" in size. How are yaw & pitch characterized at the 0.5" level?
- Answer: (a) Centroids give subpixel resolution, and (b) typically track 24 stars per frame.



Four Questions for the ST5000 Balloon Test Flight

- Would the ST5000 work in the stratosphere?
- How well would the ST5000 track?
- What motion would make the ST5000 fail?
- Would the ST5000 work in daylight?



We originally scheduled a flight from Palestine, TX in August 2010 on the HASP platform (High Altitude Student Payload – <http://laspace.lsu.edu/ hasp>), but NASA had a temporary flight moratorium that would push us back to summer 2011.

HASP payload

SIP (NASA's Support Instrument Package)

• Ballast



Near Space Corporation ST5000 Balloon Test Flight

- To fly without delay, we contracted with Near Space Corporation (Tillamook, OR) to integrate and fly our payload.
- Payload Integration at Near Space Corp.
- Used two iridium modems for command uploads only (one for the ST5000, another for the NSC command module.
- VHF data downlink
- Finally launched from Madras, Oregon on May 5, 2011.
- Balloon released at 1:30 AM, 100 minute ascent, at float past sunrise, termination near 6 AM.



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Video recorded during ascent and float phases. Considerable gondola motion - especially during ascent - but many moments of sufficient stability allowed tracking and lost-in-space solutions.







Rate of Motion (deg/sec)

Question: What are the maximum rates at which the ST5000 can track?

Answer: about 0.4 to 0.6 degrees per second.

We know this from the maximum recorded rates between pairwise orientations observed when the ST5000 was tracking.





Daytime Performance: The ST5000 stopped identifying stars during twilight, about 30 minutes before sunrise.



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PART THREE: FUTURE WORK

There are two ways to improve daytime performance:

- Infrared detector (less sky background)
- Finer platescale (less background per pixel). Requires a longer focal length and a larger detector.



DAYSTAR (University of Colorado)

We are supporting a CU Aerospace Senior Capstone project to build a new front end for the ST5000 (new optics & detector).

- Optics: focal length = 133 mm, F/1.3 telescope. Two internal stops to decrease scattered background, plus a baffle on the end.
- Detector: a 2500 x 2000 pixel sCMOS detector with low read noise (less than 2 e-).
 Full frame read out at 30 FPS.

Jed Diller Kevin Dinkel Zach Dischner Aaron Holt Tyler Murphy Sara Schuette Michael Skeen Nick Truesdale Andrew Zizzi



DAYSTAR (University of Colorado)





A recent picture of the sCMOS sensor undergoing testing at CU.

The finer platescale will improve the rms tracking to 0.1" (at night) and 1" (during day). Same FOV as the current ST5000, same ST5000 back-end. First tests expected in early April.

Jed Diller Kevin Dinkel Zach Dischner

Aaron Holt Tyler Murphy Sara Schuette Michael Skeen Nick Truesdale Andrew Zizzi



Recent Developments in NASA's Balloon Program

- Development of SUPER-PRESSURE balloons for extended day/night missions. NASA's 18 MCF balloon capable of lifting 5000 lbs suspended payload.
- 40 and 50 day missions have already flown. You can now propose for 100 DAY MISSIONS (circumglobal missions launched from New Zealand). Example: GUSSTO mission.
- The Wallops Arc-Second Pointing System (WASP) was successfully flown on a balloon on Oct 9, 2011.A 1500 lb, 24 ft mock telescope was held on target with rms yaw and pitch pointing errors of 0.25".
- Near Space Corporation is now a NASA launch provider. No cost to PIs through the Flight Opportunities program (like the NASA/BPO flights). NSC has the ability to support rapid launches of smaller payloads.



Summary

- Will the ST5000 work on a stratospheric balloon? **YES.**
- How will it perform? 0.5" (rms)
- What rates of motion will make it fail? 0.5° s⁻¹
- What sky brightnesses will make it fail? **30 min before sunrise**
- FUTURE WORK includes DAYSTAR for use on NASA's ultra-long duration balloons. We will try to piggyback a DAYSTAR test on the next WASP engineering flight.





A Strawman Planetary Payload

From Young et al. 2011, "Venus Stratoscope: A Balloon-Borne Campaign to Study Venus' Atmosphere and Surface," submitted to NASA/Planetary Astronomy.





- One meter aperture fits in WASP.
- Two cameras, IR (F/12) and CCD (F/30)
- Fast exposures (0.2 s), no Fine Steering Mirror. Over 90% of images should be diffraction limited (because of WASP).
- Estimated flight duration: 3 8 weeks.
 Estimated Cost (build, fly, recover): \$3.85M



A Strawman Planetary Payload



Venus from a stratospheric telescope:

- Frequent opportunities (every 19 months).
- Imaging at 1.74 µm shows lower cloud deck, shorter wavelengths determine surface emissivities.
- Wavelengths within 2.25 2.45 µm window: trace gas retrievals, determine cloud properties.
- Jan 2014: Venus will be continuously visible from Antarctica. Fits with NASA's existing Antarctica program.
- Resolution at 1.74 µm is 0.44" for a 1-m telescope (88 km on Venus). Cloud tracking to recover winds with 2 m/s precision.



More High-Resolution Targets

Fig 1. Spatially resolved spectroscopy of Pluto & Charon from the VLT/NACO. Protopapa et al. 2010

Fig 3. HST/HRC discovery of the binary centaur (42355) 2002 CR₄₆ with 300 sec of ontarget integration. Noll et al. 2006.

Fig. 2 Neptune (HST) in the 0.619 µm filter of WFPC2 in 1994, 1997, 2001-2002 and 2004-2007 (left to right).

Some Broad Science Goals:

- Understanding basic weather on giant planets: long-term monitoring of the giant planets at wavelengths shortward of 1.0 µm.
- Clean PSFs: finding & observing faint objects, often next to bright objects. Goals include the understanding of formation scenarios (asteroids, centaurs, TNOs, satellites and comets), ring/ satellite systems, separate spectroscopy of close binaries, and discovery of faint NEOs.
- Astrometry: vastly improved occultation predictions (e.g., for TNOs), NEO orbit determination.

Balloon Science Opportunites: A simple I-m telescope in the stratosphere has a 0.12" diffraction limit, outperforming every telescope shortward of I µm except for HST.

Cost-Effectiveness of Balloon-borne Missions

Balloon launches are less expensive than spacecraft launches – our comprehensive cost for a balloon flight from NSC (payload integration, launch, flight operation and payload recovery: \$70K).

Balloon missions are less expensive than GROUND-BASED operations. Example: one night on a Keck telescope was valued at \$107K in a recent NSF/TSIP proposal. Balloons can achieve science unavailable to ground-based sites (diffraction-limited resolution at visible & UV wavelengths, plus sensitivity to wavelengths normally obscured by telluric absorption).

Consider a \$4M balloon mission that flies for 40 days (equivalent to 40 nights): cost from Keck is MORE (\$8.5M) than the balloon mission. And if you can re-use the balloon payload, the cost-advantage is even more dramatic

